



SURVIVE AND OPERATE THROUGH THE LUNAR NIGHT WORKSHOP

NOVEMBER 13, 2018
COLUMBIA, MARYLAND

Batteries and Energy Storage Power Panel Discussion

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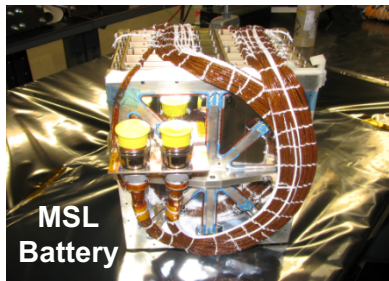
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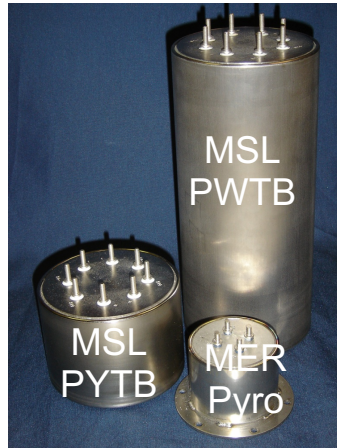
Pasadena, CA



Most NASA Missions Require Batteries



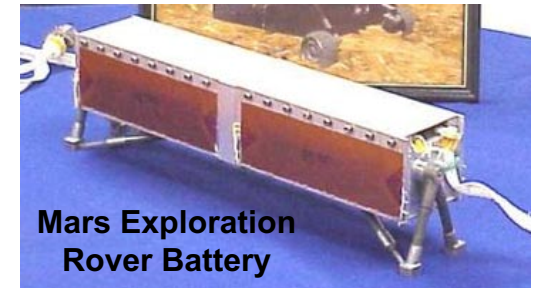
MSL Battery



Thermal Batteries



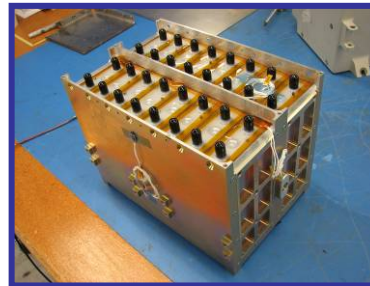
Primary MER Battery



Mars Exploration Rover Battery



Prismatic Li-ion Cells



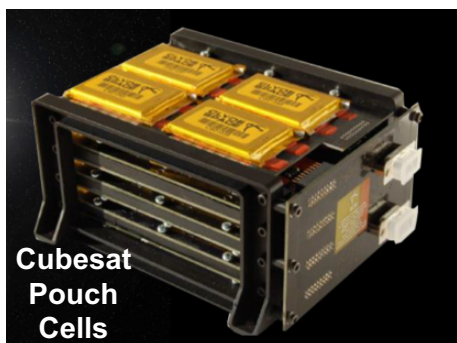
Li-ion cell battery housing



DS-2 ultra-low temperature primary cells

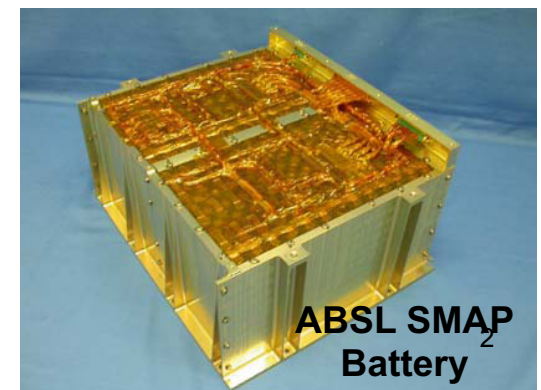


Low temperature cylindrical cells



Cubesat Pouch Cells

As a primary power source, to support load leveling or for operations involving eclipses/night operations



ABSL SMAP Battery

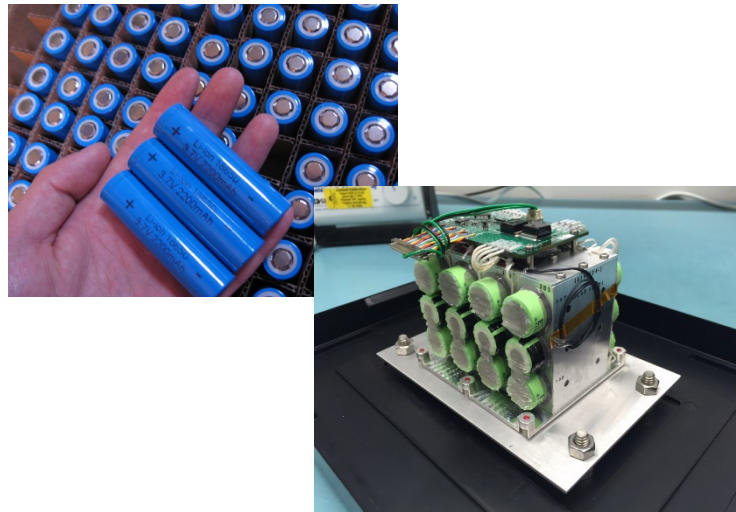


Main Cell Types Used For NASA Missions

- **Battery is comprised of multiple cells**
 - Add cells in series to increase voltage
 - Add cells in parallel to increase capacity
- **Lithium-ion cell types used for planetary exploration**
 - Large Prismatic Cells (Yardney, GS Yuasha): 5 to 190 Amp-hour capacity
 - Small Cylindrical Cells (Sony, Panasonic): 2-3 Amp-hour capacity
- **Primary Cells are typically D or larger size**
 - D or DD Cylindrical Cells (Saft, EaglePicher): 10-20 Amp-hour capacity



**Rechargeable Li-ion
Prismatic Cells**



**Rechargeable Li-ion
Cylindrical Cells**



**Primary Li
Cylindrical Cells₃**



Battery Design Approach Starts with System Analysis

- What loads are involved and what power levels are required?
- How much energy is required per cycle?
- What is the operating environment?
- What options are available for thermal management?
 - Low temperature performance (heaters, radioisotope heaters)
 - High temperature resilience (radiators, louvers)



Environment, Thermal Management Dictate Approach

Missions where adequate thermal management is available	Missions where wider operating temperatures provide system benefits	Missions where wide temperature operation is critical
<p><u>Examples</u></p> <ul style="list-style-type: none"> • RTG heat available • Avionics waste heat available • Inherent heat generation from cells based on cell chemistry 	<p><u>Examples</u></p> <ul style="list-style-type: none"> • System design benefits from wider temperature limits • Some non-resistive heat available (RHUs) 	<p><u>Examples</u></p> <ul style="list-style-type: none"> • Small augmentation missions • Very small mass, volume budgets • Little room, power for heaters, or RHUs are cost prohibitive
<div data-bbox="98 564 560 835" data-label="Image"> <p>MSL</p> </div> <div data-bbox="44 853 604 1159" data-label="Image"> <p>Europa Lander Concept</p> </div> <p>High specific energy and long life takes precedence</p>	<div data-bbox="724 559 1222 836" data-label="Image"> <p>Huygens Probe</p> </div> <div data-bbox="705 842 1250 1142" data-label="Image"> <p>InSight</p> </div> <p>Design of electrolytes has largest influence on low temperature performance and high temperature resilience</p>	<div data-bbox="1464 555 1761 792" data-label="Image"> <p>DS-2 Impactor</p> </div> <div data-bbox="1433 801 1798 1039" data-label="Image"> <p>PUFFER Rover</p> </div> <div data-bbox="1464 1043 1771 1289" data-label="Image"> <p>Orbiting Sphere (MSR)</p> </div>

Decreasing thermal management options and increasing battery customization